NEUTRONICS MODELING AND SIMULATION OF SHARP FOR FAST REACTOR ANALYSIS

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This paper presents the neutronics modeling capabilities of the fast reactor simulation system SHARP, which ANL is developing as part of the U.S. DOE's NEAMS program. We discuss the three transport solvers (PN2ND, SN2ND, and MOCFE) implemented in the UNIC code along with the multigroup cross section generation code MC²-3. We describe the solution methods and modeling capabilities, and discuss the improvement needs for each solver, focusing on massively parallel computation. We present the performance test results against various benchmark problems and ZPR-6 and ZPPR critical experiments. We also discuss weak and strong scalability results for the SN2ND solver on the ZPR-6 critical assembly benchmarks.

KEYWORDS : Neutron Transport, Simulation, Multigroup Cross Section, Fast Reactor

1. INTRODUCTION

Reactor design is an iterative process among various disciplines such as reactor physics, thermal fluid dynamics, mechanics, reactor and plant dynamics, fuel behavior, and instrumentation and control, and requires a wide range of modeling and simulation tools that can accurately predict key system performance and safety characteristics. Most code systems for fast reactor design were initiated more than twenty years ago and designed to accommodate the computing resources, tools and methods available at that time, although considerable relevant advances of methods and codes have been made more recently. Improvements to the available tools are thus needed to design future reactor systems that must meet performance, safety and economic goals higher than the current ones. The uncertainties and biases in the various areas of reactor design activities need to be reduced by enhanced prediction capabilities. These improvements are important to minimize the costly and lengthy procedures of building multiple representative mockup experiments to confirm the predictions.

The challenge in neutronics analysis is to efficiently generate solutions to the Boltzmann equation by taking into account the geometric complexity and complicated energy dependence of nuclear data. The Monte Carlo method can represent these details, but needs sufficiently low statistical uncertainty, reliable variance estimates and

uncertainty propagation. Computing resource requirements still remain unmanageable for many types of routine design analyses, including accurate estimations of detailed pin-by-pin power distributions, effects of small perturbations, thermal feedback, error propagation via fuel depletion, and transient analysis. Thus, its use is typically limited to steady-state reference solutions without thermal feedback. As a result, the current design tools rely heavily upon deterministic methods based on various approximations and sophisticated multistep procedures [1-3]. Detailed energy variable treatments are done only at the pin cell level (or using a homogeneous mixture of fuel, coolant, and structure materials) with approximate boundary conditions. Then, by performing a series of subdomain calculations with a larger problem domain but fewer modeling details, space and energy condensed parameters are defined and tabulated for global core calculations. Global analyses are performed for three-dimensional models composed of homogenized regions with low-order approximations of the Boltzmann equation (e.g., diffusion approximation). Detailed pin-by-pin information is then recovered by reconstruction (de-homogenization) methods.

In order to explore a broad range of design space and incorporate innovative design features, it is necessary to minimize approximations and increase the modeling capability from first principles. Significantly improved capabilities to simulate multiphysics phenomena are also

required to facilitate global design optimizations. For example, in fast reactors where the coolant temperature and its rise across the core are much larger than in light water reactors, the reactivity feedback due to radial expansion of the core is one of the most important reactivity feedback phenomena [4,5]. In order to reliably evaluate this reactivity feedback effect, the geometrical deformation of assembly ducts and core structures needs to be determined accurately by consistently coupled neutronics, thermal fluids and thermo-structural analyses. With the current neutronics analysis tools based on homogenized assembly models and structured meshes, however, structural deformations across a mesh interface cannot be modeled accurately. The reactivity effect is thus not properly accounted for.

As part of the U.S. Department of Energy's Global Nuclear Energy Partnership (GNEP) program, an effort was launched in April 2007 to develop the SHARP (Simulation for High-efficiency Advanced Reactor Prototyping) [6] fast reactor simulation system. The initial plan was to develop and implement revolutionary, high-fidelity reactor simulation capabilities for sodium-cooled fast reactor (SFR) design and safety analysis, with an emphasis on highly scalable simulation capabilities that can harness the power of current generation high performance computing (HPC) capacity in preparation for future high performance desktop machines. The initial plan focused on completing full-core transient simulations using the largest computing resources available (beyond exascale) at the end of the first decade of development. In October 2008, this effort was rearranged into the Reactor IPSC (Integrated Performance and Safety Code) of the Nuclear Energy Advanced Modeling and Simulation (NEAMS) program of the U.S. DOE, with emphasis on expanding its applicability from the initially targeted SFR to other envisioned advanced systems.

Under this Reactor IPSC program, the high-fidelity deterministic neutron transport code UNIC is being developed [7-9]. A long-term goal is to replace the multi-step averaging approximations by progressively applying more accurate treatments of the entire space-angle-energy phase space with sufficiently fine-grained levels of discretization. Given that high-fidelity transport calculations are not required in all areas of reactor analysis, it is also desirable to develop an analysis tool that can allow the user to start at the current level of reactor analysis and transition smoothly (i.e., with familiar input/output) to less crude homogenization approaches and eventually to fully heterogeneous descriptions. In this way, the experienced reactor analyst can choose the desired level of approximation appropriate for their computational resources and analysis goals. In order to faithfully represent the reactivity feedback caused by structural deformation, we use a conforming unstructured, finite-element mesh representation of the domain. The current mesh generation tool is CUBIT [10] while all visualization of the solution

is done with VisIt [11]. At present, three transport equation solvers have been developed: PN2ND, SN2ND, and MOCFE. PN2ND is based upon the second-order even-parity transport equation and implements spherical harmonics for the angular approximation. SN2ND is also based upon the second-order even-parity transport equation but a discrete ordinates approximation is applied for the angular approximation. These solvers are designed to use 100,000 processors or more for reasonably large problems with reduced homogenization approximations (e.g., pin-cell level). MOCFE is a method of characteristics solver for the first-order transport equation, and its targeted application is for explicit geometry, fine-group problems with the short-term purpose of cross section generation and the long-term purpose of fine-level transport calculations without homogenization. UNIC has been applied to detailed simulations of full-core mockup critical experiments on up to 295,000 processor cores. In addition to the finite-element-based solvers above, the capability of the nodal transport code VARIANT [12,13] is being rebuilt in UNIC to be applied to assembly-homogenized problems, called NODAL. NODAL is currently limited to diffusion theory and can potentially be used as one step in a multigrid preconditioner for the other solvers in UNIC.

To generate multigroup cross sections in fast reactor applications, the MC²-3 code [14] has been developed by improving the resonance self-shielding and spectrum calculation methods of MC²-2 [15] and integrating the one-dimensional cell calculation capabilities of SDX [16]. For ultrafine group spectrum calculations, to eliminate the limitations of the generalized resonance integral method of MC²-2, the resolved resonance cross sections are self shielded by numerical integrations of pointwise cross sections. The continuous slowing-down method used in the resolved resonance energy range has been replaced with the multigroup method in order to minimize the approximations involved in the slowing-down by light elements (especially hydrogen). The isotropic inelastic scattering approximation has also been removed. For the optional hyperfine group (HFG) spectrum calculation in the resolved resonance range, the isotropic source approximation has been eliminated by incorporating anisotropic scattering sources. In addition, the one-dimensional transport calculation of SDX, which was previously limited to a fine group level, has been extended to ultrafine and hyperfine group levels. The MC²-3 code is being incorporated into UNIC to alleviate the approximations used in cell calculations by the use of transport solutions for extended problem domains and to generate the cross sections consistently with the material and temperature distributions used in transport calculations.

In this paper, we report on these early efforts to develop an improved neutronics modeling capability for fast reactor analysis. The solution methods and modeling capabilities of UNIC and MC²-3 are discussed in Section 2. Initial test results for various benchmark problems and critical

experiments are presented in Section 3. The conclusions and future work are summarized in Section 4.

2. SOLUTION METHODS AND MOELING CAPABILITIES

The simulation of generally time-dependent, multiphysics phenomena requires solving the time-dependent Boltzmann equation along with the delayed neutron precursor equations. Representing the first-order time derivative by a finite difference approximation, however, the time-dependent Boltzmann equation can be cast into the time-independent transport equation with modified absorption and source terms. Thus, for simplicity, we will discuss our methods using the steady-state neutron transport equation, which can be written in standard notation as:

$$\begin{aligned} \Omega \cdot \nabla \psi(r, E, \Omega) + \Sigma_t(r, E) \psi(r, E, \Omega) \\ = S_s(r, E, \Omega) + \frac{1}{4\pi} S_f(r, E) + S_{ex}(r, E, \Omega) \end{aligned} \quad (1)$$

where S_s , S_f , and S_{ex} are the scattering, fission and external neutron sources, respectively. The scattering and fission sources are given by

$$\begin{aligned} S_s(r, E, \Omega) = \sum_i N_i(r) \int_{E'} dE' \int_{4\pi} d\Omega' \sigma_s^i \\ (E' \rightarrow E, \Omega \cdot \Omega') \psi(r, E', \Omega') \end{aligned} \quad (2)$$

$$\begin{aligned} S_f(r, E) = \sum_i N_i(r) \int_0^\infty dE' \chi^i(E' \rightarrow E) v^i \\ (E') \sigma_f^i(E') \int_{4\pi} d\Omega' \psi(r, E', \Omega') \end{aligned} \quad (3)$$

where the index i is used to denote the isotope dependency of scattering and fission properties. The angular distribution of scattered neutrons is typically represented as a Legendre polynomial series and thus the scattering source can be written as:

$$\begin{aligned} S_s(r, E, \Omega) = \sum_i N_i(r) \sum_{l=0}^L \sum_{k=-l}^l \frac{2l+1}{4\pi} Y_{lk}(\Omega) \\ \int_{E'} dE' \sigma_{sl}^i(E' \rightarrow E) \int_{4\pi} d\Omega' Y_{lk}^*(\Omega') \psi(r, E', \Omega') \end{aligned} \quad (4)$$

Here σ_{sl}^i is the l -th Legendre moment of the scattering kernel of isotope i , Y_{lk} is the spherical harmonics, and Y_{lk}^* is the complex conjugate of Y_{lk} . When there is no external

source, Eq. (1) is generally cast into the form of an eigenvalue problem by dividing the fission source by the multiplication factor.

The transport equation solvers of UNIC solve the multigroup form of Eq. (1), which can be written as

$$\begin{aligned} \Omega \cdot \nabla \psi_g(r, \Omega) + \Sigma_{tg}(r) \psi_g(r, \Omega) = Q_g(r, \Omega), \\ g = 1, \dots, G \end{aligned} \quad (5)$$

For the criticality problem, the group source Q_g can be written in terms of macroscopic cross sections as

$$\begin{aligned} Q_g(r, \Omega) = \sum_{g'=1}^G \sum_{l=0}^L \sum_{k=-l}^l \frac{2l+1}{4\pi} Y_{lk}(\Omega) \Sigma_{sl}^{g' \rightarrow g}(r) \\ \int_{4\pi} d\Omega' Y_{lk}^*(\Omega') \psi_{g'}(r, \Omega') \\ + \frac{1}{4\pi k} \sum_{g'=1}^G \chi_{g' \rightarrow g} v \Sigma_{fg'}(r) \int_{4\pi} d\Omega' \psi_{g'}(r, \Omega') \end{aligned} \quad (6)$$

where k is the system eigenvalue known as the multiplication factor. To facilitate the discussion below, we separate out the within-group scattering source and rewrite Eq. (5) as

$$\begin{aligned} \Omega \cdot \nabla \psi_g(r, \Omega) + \Sigma_{tg}(r) \psi_g(r, \Omega) = W_g(r, \Omega) + S_g(r, \Omega), \\ g = 1, \dots, G \end{aligned} \quad (7)$$

Here W_g is the within-group scattering source, and S_g includes the out-of-group scattering source and the fission neutron sources. While we are developing a Krylov subspace solver over the entire space-angle-energy system for the SN2ND and MOCFE, the bulk of the work thus far has focused on the preconditioner which requires the approximate solution of Eq. (7).

This section presents an overview of the three transport solvers implemented in UNIC for the system of equations in Eq. (7) and the methods of MC²-3 to determine the multigroup cross sections included in Eq. (7).

2.1 PN2ND and SN2ND Solvers

As mentioned in the introduction, both the PN2ND and SN2ND solvers are based upon the second-order even-parity transport equation. These second-order methodologies that implement continuous, spatial finite-element approximations permit the parallelization successes

achieved in other fields with the Poisson equation (e.g., parallel conjugate gradient methods) to be used directly in UNIC. To derive the second-order methods, the angular flux in Eq. (7) is first split into even- and odd-parity components with respect to the angular variable as

$$\psi_g(r, \Omega) = \psi_g^+(r, \Omega) + \psi_g^-(r, \Omega) \quad (8)$$

where the superscript + denotes even parity and the superscript - denotes odd parity. Evaluating Eq. (7) for Ω and $-\Omega$, and adding and subtracting the resulting equations [17], the coupled, first-order even-parity and odd-parity equations can be obtained using Eqs. (7) and (8) as:

$$\Omega \cdot \nabla \psi_g^\pm(r, \Omega) + \Sigma_{tg}(r) \psi_g^\pm(r, \Omega) = W_g^\pm(r, \Omega) + S_g^\pm(r, \Omega) \quad (9)$$

The odd-parity equation can be solved for the odd-parity flux as:

$$\begin{aligned} \psi_g^-(r, \Omega) = & \frac{1}{\Sigma_{tg}(r)} [-\Omega \cdot \nabla \psi_g^+(r, \Omega) \\ & + W_g^-(r, \Omega) + S_g^-(r, \Omega)] \end{aligned} \quad (10)$$

Substituting Eq. (10) into the even-parity equation, the second-order even-parity transport equation can be obtained as:

$$\begin{aligned} & -\Omega \cdot \nabla \frac{1}{\Sigma_{tg}(r)} \Omega \cdot \nabla \psi_g^+(r, \Omega) + \Sigma_{tg}(r) \psi_g^+(r, \Omega) \\ & = W_g^+ + S_g^+ - \Omega \cdot \nabla \frac{1}{\Sigma_{tg}(r)} (W_g^- + S_g^-) \end{aligned} \quad (11)$$

This differential equation is accompanied by the boundary conditions that constrain the angular flux along the boundary of the problem domain and at element interfaces.

In the PN2ND solver, the even- and odd-parity angular fluxes in Eq. (11) are respectively expanded in terms of even- and odd-parity spherical harmonic trial functions as:

$$\psi_g^\pm(r, \Omega) = \sum_{l=0,2,\dots,N-1} \sum_{k=-l}^l Y_{lk}(\Omega) \psi_{gk}^\pm(r) \quad (12)$$

$$\psi_g^-(r, \Omega) = \sum_{l=1,3,\dots,N} \sum_{k=-l}^l Y_{lk}(\Omega) \psi_{gk}^-(r) \quad (13)$$

where N is the angular expansion order. Figure 1 graphically displays some selected angular trial functions. The SN2ND solver uses a discrete ordinates cubature to represent the angular variable. Because a spherical harmonics-based scattering kernel is used, the cubature on the unit sphere that results from a discrete set of directions Ω_n is required to satisfy the following two relationships:

$$\sum_n Y_{lk}(\Omega_n) w_n = \delta_{l0} \delta_{k0} \quad (14)$$

$$\psi_{gk}^+(r) = \sum_n Y_{lk}(\Omega_n) w_n \psi_g^+(r, \Omega_n) \quad (15)$$

where Y_{lk} is an even-parity spherical harmonic function and w_n 's are the weights associated with the cubature. The $\psi_{gk}^+(r)$ moments are identical to those obtained using the spherical harmonic method. Currently, Carlson's level

Fig. 1. Graphical Depictions of Spherical Harmonic Functions used in PN2ND

symmetric cubature [18], the Legendre-Tchebychev product cubature [19], the Thurgood cubature [20], the Lebedev-Laikov cubature [21], the COBE sky cube-based cubature [22], and the Tegmark icosahedron-based cubature [23] have all been implemented into UNIC. Sample cubatures are shown in Fig. 2, where the black dots indicate the cubature abscissa intersections with the unit sphere and the coloring indicates the distribution of weights associated with the cubature.

The angular approaches are combined with a spatial finite element method [24, 25]. The problem domain is subdivided into a series of unstructured geometric elements that do not overlap, and the border of each element is defined using a series of spatial vertices which incorporate a distinct numbering scheme to facilitate computer implementations. For each type of finite element, a set of polynomial basis functions is defined within each element to satisfy the patch test. Lagrangian, serendipity, and Gauss-Lobatto polynomial trial functions are available in UNIC on six primary geometries: bar, triangle, quadrilateral, tetrahedron, triangular prism, and hexahedron. The Lagrangian, serendipity, and Gauss-Lobatto elements are available up to the ninth, second, and sixth orders, respectively. UNIC typically assumes an isoparametric geometry input, but it can easily be used on subparametric and superparametric geometric input definitions. The vertex layout for the quadratic serendipity and cubic Lagrange elements in the standard coordinate system are

Fig. 2. Examples of Discrete Ordinates Cubatures Available in SN2ND

shown in Fig. 3.

Both PN2ND and SN2ND assume a continuous finite element formulation for the even-parity transport equation and a discontinuous finite element formulation of the odd-parity transport equation. This reduces Eq. (11) to a linear algebra equation for the even-parity flux ψ_g^e of the form:

$$A_g \psi_g^+ = B_g \psi_g^+ + S_g^+ + C_g (W_g^- + S_g^-) \quad (16)$$

where B_g represents the within-group scattering matrix. Arbitrarily oriented, reflected and vacuum boundary conditions are included along with the ability to include volumetric fixed sources. The odd-parity sources of Eq. (16) are iteratively determined with the use of Eq. (10). The coefficient matrix A_g is positive definite, and thus Eq. (16) can be solved efficiently using the conjugate gradient

Fig. 3. Examples of Serendipity and Lagrange Finite Elements

method [26], when the source terms on the right hand side lagged in the iterative approach discussed below.

For PN2ND, the coefficient matrix A_g and the within-group scattering matrix B_g exhibit the coupling of $\psi_{g'l'k}^*$ with $\psi_{g'l'k}^*$ for $l' = l$ or $l - 2$, and $k' = k - 1, k$, or $k + 1$. Because these matrices have the same coupling, the within-group scattering source W_g^* is moved to the left hand side and B_g is merged with A_g in PN2ND. Reflected boundary conditions do not fundamentally change the coupling, but vacuum boundary conditions couple all angular terms together for those spatial vertices that lie along the boundary in the PN2ND coefficient matrix.

In SN2ND, no coupling occurs between the angles when no reflected boundary condition is present, and thus the A_g matrix is block-diagonal with respect to angle. When reflected boundary conditions are present, those vertices that lie along a reflected boundary have coupling in angle, which appears as a very sparse connection in A_g . When reflected boundary conditions are present, experience has shown that the angular cubature should display the underlying symmetry in the problem, but SN2ND does not impose this behavior and can use cubatures which do not have the necessary symmetry. In general, A_g matrix always remains nearly block diagonal with respect to angle because the number of vertices that lie upon the reflected surfaces of the domain is far fewer than the number that are internal to the domain and the implemented angular cubature generally matches the underlying symmetry of the domain. Unlike the A_g matrix, the within-group scattering matrix B_g is full with respect to angle. Thus Eq. (16) is solved iteratively by estimating the within-group scattering source using the most recent flux solution. While convergence of this equation is generally guaranteed, the speed of convergence strongly depends upon the ratio of scattering to collision and the amount of leakage. As such, acceleration schemes have been developed based upon the diffusion equation. For the SN2ND solver, the diffusion operator is a direct projection of the even-parity transport operator and thus the diffusion synthetic acceleration (DSA) scheme [27] is automatically fully consistent and not prone to the weakened convergence properties observed in the first-order discrete ordinates solvers.

From a parallelization point of view, $A_g - B_g$ in PN2ND and A_g in SN2ND are both scalable operations because they can both be solved via conjugate gradient (CG) methods which have been shown to perform well on small- and large-scale parallel machines. Others have reported in the literature their application of CG methods to similar equations with processor counts as high as 10,000 and with 70 to 80% efficiency. To achieve such good performance, the literature indicates that developers should create a preconditioner that contains domain decomposition with multilevel p -refinement in space in addition to multiple levels of h -refinement and/or algebraic multigrid. In UNIC, to partially address the memory issue, we employed parallelization by angle rather than p -refinement or h -refinement. Thus for the

SN2ND solver with a block diagonal connection in angle, it is not necessary to invest time defining a multigrid operation in angle because the method is immediately scalable in angle. It is expected that by incorporating the methodology that other codes have been developed to apply the CG method on up to 10,000 processors, the SN2ND solver can efficiently use over 1,000,000 processors to solve a problem with 100 discrete ordinate directions.

The within-group equations given in Eq. (16) are coupled through the source S_g and form a multigroup eigenvalue or fixed-source problem. The resulting multigroup equation is solved using a standard fission source iteration algorithm. To find the dominant eigenvalue, the power method has been used with a Tchebychev acceleration scheme. For a known fission source from the previous power iteration, another level of iteration on the multigroup scattering source is introduced because of the energy-angle coupling that occurs in the scattering operator. To decouple the group equations, a Gauss-Seidel procedure in energy is typically employed. The ultimate intent with UNIC is to implement a more general Krylov method where the current Gauss-Seidel scheme would likely serve as some part of a preconditioner where scalability in energy for time-dependent problems is assumed. At this stage, we have Eq. (16) with a known out-of-group source from the most recent fission and multigroup scattering source iterations. To solve Eq. (16) in a parallel environment, the parallel CG implementation in PETSc [28] is used, where either the incomplete Cholesky (ICC) or symmetric successive over-relaxation (SSOR) preconditioner provided by PETSc was used. Each of these levels of iteration has simple dynamic error controls to minimize effort early in the calculation.

As mentioned above, SN2ND uses a conventional iterative scheme on the within-group scattering source (source or scattering iteration). Within each scattering iteration, the entire space-angle system needs to be solved using a CG operation because reflective boundary conditions inherently couple the individual angular systems together as mentioned earlier. Finally, the CG preconditioner for each discrete ordinate direction uses a preconditioned CG operation on the full spatial domain. Thus, the solution methodology in SN2ND consists of the following layers of iteration:

Fission source (inverse power) iteration for the eigenvalue (or fixed source)			
			Gauss-Seidel iteration over energy groups
			Scattering iterations for the within-group scattering system
			Conjugate gradient over the whole space-angle system
			PETSc preconditioned conjugate gradient for each discrete ordinate

At the lowest level, a parallel SSOR-preconditioned CG methodology available in PETSc is used to solve the full spatial domain for a given angle and energy group. This

requires distributed sparse matrix storage of each symmetric positive-definite matrix. A preconditioned CG iteration is currently used for the second-lowest level, but a multigrid preconditioner is being developed to reduce sparse matrix storage requirements [29,30]. As mentioned above, DSA is used at the third-lowest level to accelerate the within-group scattering source iteration.

With regard to parallelization, we employ the generic decomposition of space, angle and energy shown in Fig. 4. In the figure, the notation “ S,A,G ” represents parallelization into S spatial segments, A angular segments, and G energy segments. The main focus is to limit the communication costs of the various source operations necessary in the within-group formulations that are fundamental to modern neutron transport solvers. In this approach, each MPI process sees four communicators: space, angle, group, and the global communicator. The advantage of this approach is that the group and angle communication do not overlap with respect to space, and thus communication in these two directions can be done simultaneously on the parallel machine. Parallel hierarchy proceeds as energy, angle, and space in descending order. For the spatial part, classic domain decomposition is used where weights are applied to the vertices of elements to balance the local work with the communication costs required to connect the domain. Figure 5 shows an example of spatial domain decomposition obtained with the MeTiS package [31]. No solver in UNIC currently allows decomposition of the energy domain, although UNIC itself has been built this way from the beginning. It is important to note that the current space-angle scheme in SN2ND is already enough to saturate the available memory and processors of the largest supercomputers available to open science research. In the current SN2ND solver, any number of angles can be distributed on a given process and it has been found that two to three angles per process works best.

2.2 MOCFE Solver

The MOCFE solver is based upon the first-order method of characteristics (MOC) [32,33] and treats both two- and three-dimensional geometries. The long-term development goal for MOCFE is to use this solver on explicit geometry problems for high-fidelity benchmark calculations. However, the computational performance of MOC is too slow to be applied for three-dimensional whole core problems, and thus the short-term goal of MOCFE is to improve the cross-section generation process. As an example, a whole core planar two-dimensional MOC capability is a focused development effort intended to improve the spectral transition of the flux as it exits the core and enters the blanket and reflector in global three-dimensional solutions.

The derivation of the MOC equations begins with the first-order transport equation given in Eq. (7). Basically, noting that the first term on the left hand side of Eq. (7) is a directional derivative of ψ_g in the direction of Ω , the

Fig. 4. Parallelization Strategy in UNIC

Fig. 5. An Example of MeTiS Decomposition for 512 Processors

solution of Eq. (7) along a trajectory (ray) parallel to Ω (i.e., a characteristic line) can be obtained analytically for a given source distribution, assumed to be piecewise constant for this derivation. Thus, by evaluating the angular fluxes along the path of trajectories that are parallel to each direction in a selected angular cubature (i.e., ray tracing), the angular and scalar fluxes throughout the geometry on a finite element grid can be determined. This leads to a discrete form of the balance equation, which can be iteratively solved by updating the scattering and fission source terms with the most recent flux solution. The details of the MOC formulation were discussed previously [34,35] and thus not reproduced here.

Unlike most MOC solvers, MOCFE is based on a finite element discretization of the spatial domain rather than a combinatorial geometry. This is done primarily to handle truly arbitrary geometries that can account for structural deformation feedback, which is not easy to achieve with typical combinatorial geometry approaches. It is important to note that for a given combinatorial geometry, one can easily construct a finite element mesh, and thus the common simplified geometrical reactor core descriptions can be handled by the same rules under which existing MOC solvers operate. The current MOCFE solver can assume all finite elements supported by UNIC are cast onto the five primary ray tracing finite elements shown in Fig. 6. For two-dimensional elements, a standard line-intersection algorithm is used, while a ray-triangle intersection algorithm [36] is used for three-dimensional elements. This latter algorithm requires breaking the surfaces of the finite elements into triangles, the minimal decomposition of which is shown for each element type in Fig. 6. A scalable back projection algorithm has been constructed to define the trajectories in such a way that every element in the domain has crossing trajectories.

Fig. 6. Lowest-order Ray Tracing Elements in the MOCFE Solver

In the original development [34], parallel versions of MOCFE in two- and three-dimensional geometries were created assuming long characteristics over the whole domain. The trajectories were partitioned for each direction, but it was assumed that all processors have access to the full spatial domain. While this approach worked on a few hundreds of processors, the parallel algorithm employed was not scalable on petascale machines. After gaining some experience in the field, it was concluded that domain decomposition with per process segmentation of the characteristic lines is essential to have a scalable algorithm on petascale machines.

A new solver methodology has been developed using a hierarchical GMRES solver of PETSc [28] over the entire space-angle-energy system and thus a parallel decomposition in space-angle-energy. In angle, a straightforward partitioning of the discrete ordinates angles is used in a manner similar to that done in SN2ND. In space, standard spatial decomposition is used, which translates to the splitting or segmentation of the trajectories. Figure 7 shows an example of domain decomposition and trajectory segmentation for a two-dimensional assembly. The left hand picture shows a fuel assembly mesh decomposed into six pieces where the colors indicate geometric heterogeneity. In the left hand picture, the lower right image only applies coloring to the elements assigned to each processor and the upper image shows an uncolored mesh matching the right hand picture. The right hand picture shows the six-processor decomposed mesh overlaid with the intersecting trajectories (parallel lines). The breaking of each trajectory along the domain surface is clearly identifiable and reentrant trajectories can be identified on all six domains.

Along the border of each processor domain, the incoming and outgoing flux solution for each broken trajectory must be stored and communicated to the connected processor(s). One typically would consider the incoming flux to be a boundary condition for any given processor while the outgoing flux would be the boundary condition for the downwind processors. Because each domain can be reentrant with any given trajectory and any given trajectory can be reentrant on multiple domains with domain decomposition, the typical approach to sequenced sweeping is not wise. In MOCFE, the linear system of equations is recast into a form compatible with a Krylov subspace solver such that the outgoing angular trajectory flux on each domain is part of the Krylov vector space. This approach entails certain memory and computational costs and is currently being studied for use as a possible petascale algorithm for MOC. Note that MOCFE can incorporate the conventional sweep methodology, but that is not the focus of the development research.

The new version of MOCFE has been executed on 65,536 cores of BlueGene/P [37] for a mesh with 2.5 million elements. The observed strong and weak scaling

Fig. 7. Examples of Mesh Decomposition and Segmentation of Crossing Trajectories

is relatively poor primarily due to load balancing issues caused by a different number of characteristic lines passing through each subdomain. This load imbalance is directly attributable to the geometric heterogeneity of the domain (fuel assemblies have more geometry than shield assemblies) and thus one can observe a strong variation in the number of characteristic lines crossing any given domain with respect to angle. Two possible ways to fix the load imbalance are being examined at this point: one is to increase the number of angles per process (maximize the azimuthal angle between the directions) and the other is to increase the number of domains per process. Research into the first option is nearly complete with the general conclusion that a minimum load imbalance of 6.0 is achievable with low-order cubatures (20-50 directions), but this is unacceptable because it strongly restricts scalability (<40%). The second option is likely better since the subdomains can be assigned based upon a combination of mesh density, element count, and geometric location. In the end, it is expected that a combination of the two will be necessary to make the load imbalance under 2.0, which is essential to obtain a scalability above 60%.

The most important research that needs to be carried out for MOCFE is determining a better preconditioner on the space-angle system along with one on the entire space-angle-energy system. As mentioned, the new MOCFE solver incorporates Krylov, specifically GMRES, on the entire space-angle system for which a diffusion synthetic acceleration system derived from the algebraic collapsing algorithm (ACA) is used as a preconditioner. This preconditioner is very cheap with respect to memory, but that also appears to severely restrict its accuracy. In this

regard, a low-order P_N system similar to the PN2ND solver will be investigated.

As a final note, this version of MOCFE is applicable to small-scale parallel machines (100s of processors) with parallelization in angle only. Most discrete ordinates solvers have good strong (>50%) and weak (>70%) scaling in angle. This should make the lattice calculations typically used in cross-section generation feasible using the existing solver scheme.

2.3 Multigroup Cross Section Generation

The generation of multigroup cross sections from evaluated nuclear data files remains one of the fundamental problems in reactor physics because of the need for accurate treatment of resonance effects. The methods used for cross-section processing must take into account both the resonance characteristics of intermediate nuclides (sodium, oxygen in the case of oxide fuel, and structural materials like iron, chromium, nickel, etc.) and the self-shielding in fissile and fertile nuclides. The rapidly varying elastic scattering cross sections of intermediate nuclides give rise to the fine structure of a typical fast reactor spectrum and require detailed treatment. The overlapping effect resulting from neighboring resonances either from the same nuclide or from different nuclides is another challenge in fast reactor analysis. Inelastic scattering also plays an important role in fast reactors, as it is the major cause of neutron slowing-down at high energies. Resonance shielding is generally treated using the Bondarenko self-shielding factor method [38-40] and the subgroup method [41-45], these being parameterizations of the resonance structure from which

effective cross sections can be calculated for particular compositions and temperatures. An alternative approach developed at Argonne National Laboratory (ANL) is based on detailed spectrum calculations for specific compositions and temperatures [16] by representing all the resonance structures explicitly. At present, this approach is the most widely used in the United States of America for fast reactor modeling.

In the ETOE-2/MC²-2/SDX code system [16] developed by ANL, the ETOE-2 code processes the ENDF/B data and prepares the MC²-2 libraries, and the MC²-2 and SDX codes perform spectrum calculations and generate group constants for given compositions and temperatures. The current ultrafine group (UFG) energy structure chosen to form the MC²-2 libraries for smooth tabulated non-resonant data is 2082 groups with constant lethargy from 14.19 MeV to 0.4 eV, but the ETOE-2 code allows any number of UFG lethargy intervals. Wide resonances are screened out and added to UFG smooth cross sections. The resonance data provided in the Reich-Moore formalism are converted into the multipole formalism [46] that preserves the general features required by the traditional resonance integral concept and the Doppler-broadening algorithm in the MC²-2 code without compromising rigor.

The MC²-2 code solves the slowing-down equation for specific compositions and temperatures with explicit representation of resonances. It solves the extended transport P₁, extended transport B₁, consistent P₁, and consistent B₁ fundamental mode UFG equations and accommodates high-order anisotropic scattering representations. Resolved and unresolved resonances are treated explicitly by the generalized resonance integral formulation based on the narrow resonance (NR) approximation including overlapping and Doppler broadening effects [47]. A fundamental mode homogeneous unit cell calculation is performed by solving the multigroup slowing-down equation above the resolved resonance energy and the continuous slowing-down equation below this range [48]. Equivalence theory is used to treat the heterogeneity effect, and isotropic approximation is used for fission, inelastic, and (n,2n) sources. An alternative hyperfine group (HFG) integral transport calculation (RABANL) is available as an option to model the resolved resonances more accurately in the low-energy ranges where the narrow resonance approximation is not valid. Under this HFG method, every resolved resonance is represented by multiple points, which allows the neutron spectrum within a HFG to be approximated as a constant. RABANL performs a homogeneous or heterogeneous (pin or slab) unit cell calculation over the resonance region using the isotropic source approximation.

The SDX code is intended to supplement what is lacking in MC²-2 for treating the detailed heterogeneity effects associated with a complex reactor system. For each heterogeneous unit cell, heterogeneous resonance cross sections are computed in an intermediate group level of

230 groups for selected isotopes in the specified plate or pin types using equivalence theory or the rigorous RABANL heterogeneous treatment. These resonance cross sections are combined with the intermediate-group library data constructed from MC²-2 ultrafine group calculations, and then an infinite slab or cylinder integral transport calculation is performed in the intermediate group level for the unit cell. The integral transport calculation is based on a modified version of the CALHET code that applies the collision probability methods developed for RABANL. Spatial self-shielding factors and cell-averaged group cross sections are calculated.

In recent studies with the ENDF/B-VII.0 data [49,50], MC²-2 revealed some discrepancies. For example, the increased importance of resolved resonances in the ENDF/B-VII.0 data due to the extended upper energy cutoff and significantly increased number of resolved resonances required the use of RABANL for rigorous treatment of resolved resonances. However, its use is limited to the relatively low energy range where the isotropic source approximation is valid. Therefore, in order to generate the more accurate multigroup cross sections required for high-fidelity simulation with UNIC, MC²-3 has been developed by integrating the capabilities of MC²-2 and SDX and by eliminating various limitations and approximations of the current methods as discussed below.

The generalized resonance integral method of MC²-2 for calculating self-shielded UFG cross sections is based on the analytical integration of resonances over the entire energy range rather than each UFG interval. The approximate integration interval introduces the so-called resonance tail effect, and thus wide resonances are screened out and added to UFG smooth cross sections. The resonance tail effect becomes more pronounced as the number of energy groups in the target library increases, and thus the analytic resonance integral method becomes less attractive for the high-fidelity simulation with UNIC where the number of energy groups is expected to increase with increases in computing power. Therefore, a new approach based on the numerical integration of pointwise cross sections with the narrow resonance approximation has been introduced in MC²-3. Doppler broadened pointwise cross sections are reconstructed within the code using the resonance parameters. The NR fluxes are obtained using the pointwise total cross sections of a specified composition. The self-shielded UFG cross sections are determined using numerical integration over the exact energy interval of each UFG as:

$$\bar{\sigma}_x^i = \int_{\Delta u} \frac{\sigma_x^i(u)}{\Sigma_t(u)} du / \int_{\Delta u} \frac{1}{\Sigma_t(u)} du \quad (17)$$

where $\sigma_x^i(u)$ is the pointwise cross section of reaction type x for isotope i in the lethargy interval Δu and $\Sigma_t(u)$ is

the macroscopic total cross section of the composition. In a similar way, the l -th order self-shielded cross sections are calculated by weighting with the l -th order flux moment approximated as $1/[\Sigma_l(u)]^{l+1}$.

The continuous slowing-down method of MC²-2 used for the UFG spectrum calculation in the resolved resonance energy range becomes more approximate when light elements (especially hydrogen) are included. In order to minimize this approximate nature and to simplify the program structure, the continuous slowing-down method has been replaced with the multigroup method. Consequently, the UFG spectrum is calculated by solving the multigroup P_1 equations for the entire energy range with an extended transport approximation [15,17] up to P_9 . The isotropic approximation of inelastic scattering and (n,2n) reactions has also been removed by introducing anisotropic scattering matrices. In addition, the isotopic fission spectra depending on incident neutron energy are available, and the resonance-like cross sections of the intermediate mass nuclides such as iron, nickel, and chromium above the resonance energy range are self-shielded using the NR approximation.

For the HFG spectrum calculation in the resolved resonance range, the isotropic source approximation is eliminated by incorporating anisotropic scattering sources. This allows increasing the upper energy limit of HFG calculation up to the top of the resolved resonance region of intermediate mass nuclides (~1 MeV) where the isotropic source approximation is not valid. As mentioned above, the HFG structure is constructed in such a way that the neutron spectrum within a HFG can be approximated to be constant. Thus, if N is the scattering order in the center-of-mass system (CMS) given in the ENDF/B data, the scattering transfer cross section of order l from the source group g to the sink group g' in the laboratory system (LS) is calculated using

$$\sigma_{sl}(g \rightarrow g') \cong \int_{u_{g'-1}^*}^{u_{g'}^*} du' \int_{u_{g-1}^*}^{u_g^*} du \frac{\sigma_s(u) e^{-(u'-u)/2} P_l[\mu_0(u, u')]}{(1-\alpha)} \quad (18)$$

$$\sum_{n=1}^N (2n+1) f_n(u) P_n[\mu_c(u, u')]$$

where f_n is the n -th order Legendre expansion coefficient in CMS, $u_{g'}^*$ and u_{g-1}^* are the reachable energy boundaries, and $\alpha = (A-1)^2 / (A+1)^2$ with A being the atomic mass. The cosines of the scattering angles in LS and CMS are given by

$$\mu_c(u, u') = \frac{1}{1-\alpha} [2e^{-(u'-u)/2} - (1+\alpha)] \quad (19)$$

$$\mu_s(u, u') = \frac{1}{2} [(A+1)e^{-(u'-u)/2} - (A-1)e^{-(u-u')/2}] \quad (20)$$

The heterogeneous cell calculation capability of SDX for slab and cylindrical unit cells has also been integrated into MC²-3 by extending the spatial transport calculation from an intermediate group level to the ultrafine group level. Either the collision probability method (CPM) or the method of characteristics (MOC) can be used. The UFG cross sections of each region are determined using the equivalence theory. That is, the UFG cross sections are self shielded using the composition-dependent potential scattering cross section, augmented by the geometry-dependent escape cross section. Region-wise or cell-average broad-group cross sections are produced. A HFG transport calculation capability with an anisotropic scattering source representation is currently being implemented. The MC²-3 code is also being built to use the MOCFE solver of UNIC so that the approximations imposed by the simplified one-dimensional cell calculations are reduced and the material and temperature distributions that appear in whole-core transport calculations are correctly incorporated.

3. TEST RESULTS

To test the performance of MC²-3 and UNIC, we analyzed several benchmark problems and fast critical experiments. The experiments analyzed include eight critical assemblies [51] of Los Alamos National Laboratory (LANL), ZPR-6 Assemblies 6A (ZPR-6/6A) and 7 (ZPR-6/7) [51,52], three loadings of ZPPR-15A [53], six phases of ZPPR-21 [51], and three BFS assemblies [52,54]. Using the ENDF/B-VII.0 data [55], isotopic multigroup cross sections were generated for each composition using MC²-3. The MC²-3 and UNIC results were compared with the reference Monte Carlo solutions obtained with MCNP5 [56] or VIM [57]. For detailed as-built models, the results were also compared against the experimental results.

The eight LANL critical assemblies include Godiva, Bigten, Flattop-25, Flattop-Pu, Flattop-23, Jezebel, Jezebel-Pu, Jezebel-23, three of which—Flattop-Pu, Jezebel, and Jezebel-Pu—contain plutonium fuel. The core regions of ZPR-6/6A and ZPR-6/7 are composed of uranium and plutonium fuels, respectively. Of the six phases of ZPPR-21, the first phase, ZPPR-21A, contains only plutonium fuel to take advantage of the inherent neutron source of Pu-240. The enriched uranium is then progressively substituted for plutonium in Phases B through E so that the final phase, ZPPR-21F, has only uranium fuel loading, which requires an external neutron source to reach criticality. Three loading configurations of the ZPPR-15A

experiments tested in this study are the initial criticality configuration (loading 15), a reference configuration for sodium void worth measurement (loading 16), and a configuration with an 18-inch sodium void in part of the inner core (loading 20). Among the three BFS assemblies, the BFS-75-1 assembly consists of two enriched uranium fuel zones and two composite radial blankets (U-238 metal and UO_2), while BFS-73-1 has only a single enriched uranium fuel zone and one radial blanket (UO_2). The difference in composition between the two assemblies is that zirconium metal disks are included in the unit fuel cell of BFS-75-1 but not in that of BFS-73-1. A unit fuel cell of the BFS-75-1 critical assembly is more heterogeneous in configuration than that of the BFS-73-1 since a highly enriched uranium disk is placed between medium enriched metal uranium disks. Similar to BFS-73-1, the BFS-55-1 assembly has a single enriched uranium fuel zone and one radial blanket, containing plutonium disks in the unit fuel cell unlike the other two assemblies.

3.1 Homogeneous Medium and Unit Cell Calculations

Using the improved methods of MC²-3, the previously observed discrepancies of MC²-2 with the ENDF/B-VII.0 data and with the increased number of broad groups were first examined by performing homogeneous medium calculations. The neutron spectra were calculated by solving three different slowing-down equations: the UFG equation with anisotropic scattering sources, the HFG equation with isotropic (in CMS) sources as in RABANL, and the HFG equation with anisotropic sources. For the UFG calculation, the self-shielded cross sections were obtained by the numerical integration of pointwise cross sections with the narrow resonance approximation as discussed earlier. Figure 8 compares the UFG spectra obtained from the three slowing-down calculations for the inner core composition of ZPR-6/6A. The HFG calculations were applied to the energy range below 1 MeV because the highest resolved resonance energy of Cr-52 included in the composition is 980 keV in the ENDF/B-VII.0 data. As can be seen, the UFG spectrum agrees well with the reference HFG spectrum obtained with an anisotropic source. This result indicates that the numerical integration of pointwise cross sections with the narrow resonance approximation is sufficiently accurate for self-shielding the UFG cross sections. Thus, it can be concluded that the required use of RABANL in MC²-2 calculations with the ENDF/B-VII.0 data is largely due to the increased resonance tail effects of the analytic resonance integral method. This conclusion is also supported by the observation that without the RABANL hyperfine group calculation, MC²-2 produces large errors in UFG total cross sections even at the infinite-dilute condition, whereas MC²-3 generates accurate infinite-dilute UFG cross sections.

It can also be seen from Fig. 8 that the HFG spectrum

obtained with isotropic (in CMS) sources is noticeably different from the reference spectrum above 100 keV because of the increased importance of anisotropic scattering due to p-, d-, and f-wave scattering. This result suggests that in this energy range, anisotropic scattering in the CMS should be accounted for. As expected, it was also observed that for low-energy groups around a relatively wide actinide resonance that spans several ultrafine groups, the UFG cross sections obtained with the NR approximation show non-negligible differences from those collapsed with the HFG flux solution. As an example, Fig. 9 compares the NR approximation flux superimposed on the UFG flux with the HFG flux around a U-238 resonance at 189.68 eV. A noticeable difference can be seen below the resonance peak. As a result, for the UFG between 186.2 to 186.7 eV, the total cross section based on the NR approximation shows ~5% deviation from that obtained with the HFG solution. As can be seen in Fig. 8, the neutron flux in this low energy range is generally very low in fast reactors, and thus the impacts of the UFG cross section errors are negligible. However, this epithermal energy range could be important for certain applications such as the steam-flooded, gas-cooled fast reactor analysis.

Table 1 compares the infinite multiplication factors obtained with MC²-3 with MCNP5 or VIM Monte Carlo solutions for the various homogenized compositions and unit cells of the ZPR-6/6A, ZPR-6/7, ZPPR-15A, and ZPPR-21 assemblies. Based on the aforementioned observations, the default UFG slowing-down option was used for MC²-3 calculations. The ZPR-6 and ZPPR fast critical assemblies are split-table machines holding lattices of stainless steel tubes with a square cross section of 5.5 cm outside dimension. These tubes are loaded with stainless steel drawers filled with plate-type unit cell loadings, as depicted

Fig. 8. Comparison of UFG Spectra Obtained with Three Different Slowing-Down Calculations

Fig.9. Comparison of NR Approximation Fluxes Superimposed on UFG Flux and HFG Flux around U-238 Resonance at 189.68 eV

in Fig. 10 for the loading 106 of ZPR-6/7. The unit cell loadings are approximately one-dimensional (1-D), but the variations in plate dimensions make them three-dimensional (3-D) as shown in Fig. 11. Since the MC²-3 unit cell calculation is based on 1-D slab geometries, the 3-D matrix tube loading was transformed to an equivalent 1-D computational model by smearing the structural materials from the periphery of the physical cell into the regions that do not include heavy elements. It can be seen that the infinite multiplication factors of MC²-3 agree well with the Monte Carlo solutions, but MC²-3 consistently underestimates the infinite multiplication factor compared to MCNP5 or VIM. It is also noted that the local heterogeneity effect is very large for ZPR drawers (e.g., 3,744 pcm for the ZPR-6/7 inner core drawer, 1,829 pcm for the ZPPR-15A inner core drawer, and 1,009 pcm for the ZPPR-15A outer core drawer), and this large local heterogeneity effect is accurately accounted for in the 1-D cell model of MC²-3.

Table 1. Infinite Multiplication Factors of Homogenized Compositions and Unit Cells

Model	Assembly	Drawer	Monte Carlo	MC ² -3 ^{a)}
Homogenized Composition	ZPR-6/6A	Inner core	1.22945 ± 0.00038	7
		Outer core	1.22482 ± 0.00048	-31
		Radial blanket	0.33513 ± 0.00043	-59
		Axial blanket	0.33215 ± 0.00048	-22
	ZPR-6/7	Inner core	1.24366 ± 0.00046	-87
		Outer core	1.23802 ± 0.00047	-128
		Radial blanket	0.33445 ± 0.00042	10
		Axial blanket	0.32975 ± 0.00043	-27
	ZPPR-15A	Inner core	1.17243 ± 0.00018	-247
		Outer core	1.65167 ± 0.00019	-284
	ZPPR-21A	Inner core	2.49784 ± 0.00013	-55
	ZPPR-21B	Inner core	2.38138 ± 0.00012	-77
	ZPPR-21C	Inner core	2.27032 ± 0.00012	-76
	ZPPR-21D	Inner core	2.16489 ± 0.00010	-104
	ZPPR-21E	Inner core	2.08656 ± 0.00011	-108
	ZPPR-21F	Inner core	2.04243 ± 0.00011	-121
1-D Unit Cell	ZPR-6/7	Inner core	1.28110 ± 0.00028	286
	ZPPR-15A	Inner core	1.19072 ± 0.00021	-54
		Outer core	1.66176 ± 0.00021	-29

^{a)} Difference from Monte Carlo solution in Δk pcm

Fig. 10. Schematics of ZPR-6/7 Loading 106

Fig. 11. Typical ZPR Drawer (top) and Cross Section of Core Unit Cell Showing Matrix and Plate-loaded Drawer of ZPR-6/7 (bottom)

3.2 Simplified Benchmark Problems

As a next step to test the performance of MC²-3, the simplified benchmark problems for the experiments discussed above were analyzed. The benchmark models in spherical or cylindrical (R-Z) geometries with homogenized compositions were used. For each distinct composition, 230-group cross sections were generated with MC²-3 using the default UFG slowing-down calculation option. The core calculations were performed using the TWODANT discrete ordinate code [27] with S₂₄ angular quadrature and P₃ or P₅ anisotropic scattering order. Table 2 compares the effective multiplication factors obtained from MC²-3/TWODANT calculations with MCNP5 or VIM Monte Carlo solutions. As can be seen, the MC²-3/TWODANT results agree well with the Monte

Table 2. Effective Multiplication Factors of Simplified Benchmark Problems with Homogenized Compositions

Assembly	Monte Carlo	MC ² -3/ TWODANT ^{a)}
GODIVA	0.99996 ± 0.00032	5
BIGTEN	0.99513 ± 0.00027	187
FLATTOP-25	1.00212 ± 0.00035	101
FLATTOP-Pu	1.00072 ± 0.00034	-134
FLATTOP-23	0.99921 ± 0.00034	-26
JEZEBEL	1.00028 ± 0.00030	-129
JEZEBEL-Pu	0.99944 ± 0.00031	-27
JEZEBEL-23	1.00007 ± 0.00031	-14
ZPR-6/6A	0.99609 ± 0.00023	-15
ZPR-6/7	0.98671 ± 0.00022	-18
ZPPR-15A L15	0.98386 ± 0.00026	-23
ZPPR-15A L16	0.98081 ± 0.00028	-11
ZPPR-15A L20	0.98413 ± 0.00025	-46
ZPPR-21A	0.99869 ± 0.00020	-133
ZPPR-21B	0.99293 ± 0.00020	70
ZPPR-21C	0.99923 ± 0.00018	-90
ZPPR-21D	1.00345 ± 0.00020	141
ZPPR-21E	1.00485 ± 0.00020	49
ZPPR-21F	1.00612 ± 0.00020	-123
BFS-55-1	0.97511 ± 0.00016	197
BFS-73-1	0.99553 ± 0.00017	149
BFS-75-1	0.99349 ± 0.00017	139

^{a)} Difference from Monte Carlo solution in Δk pcm

Carlo solutions: within ~ 200 pcm for the LANL critical assemblies and three BFS experiments; within ~ 150 pcm for six loadings of the ZPPR-21; within ~ 50 pcm for three loadings of ZPPR-15A; and within ~ 20 pcm for the ZPR-6/6A and ZPR-6/7. No obvious trend is observed in the differences.

The differences from Monte Carlo solutions of the TWODANT multiplication factors obtained with MC²-2 and MC²-3 cross sections are also compared in Fig. 12. The figure shows that the MC²-3 cross sections improve the multiplication factors significantly. In particular, the solutions for LANL assemblies improved markedly, mainly due to the new capability of treating anisotropic inelastic scattering.

Fission reaction rates of major isotopes for six LANL critical assemblies were calculated at the center of the assemblies, and compared with the measured values and MCNP results. As shown in Table 3, the fission reaction rate ratios of U-233, Np-237, U-238, and Pu-239 to U-235 calculated with MC²-3/TWODANT agree very well with those of MCNP, although all the calculated values with ENDF/B-VII.0 are smaller than the measurements by up to $\sim 4\%$. These results indicate that the deviations from the measurements are mostly due to the ENDF/B-VII.0 data itself rather than the multigroup cross sections generation method of MC²-3.

3.3 Three-Dimensional Homogenized Core Benchmarks

The three solver modules of UNIC were first tested using three-dimensional homogenized benchmark problems with given sets of group cross sections. One such series

Fig. 12. Differences from Monte Carlo Solutions of TWODANT Multiplication Factors Obtained with MC²-2 and MC²-3 Cross Sections

of benchmarks applicable to both thermal and fast reactors was proposed by Takeda [58]. The first three benchmarks are Cartesian, PWR-like thermal reactors while the fourth is a hexagonal configuration representative of a fast reactor. The original purpose of the benchmark was to compute the control rod worth. The first and fourth benchmarks were solved with UNIC. The first benchmark proposed is a small cubical reactor 50 cm on a side. The geometry has 1/4 symmetry in the radial plane and 1/8 symmetry overall. The fourth benchmark is a medium-sized fast reactor 105 cm tall with an assembly pitch of 12.99038 cm and a full 8-ring geometry definition. The geometry has 1/12 symmetry in the radial plane. For the first and fourth benchmarks, two- and four-group macroscopic cross sections were respectively provided for each region in the problem in addition to a reference Monte Carlo solution.

Two sets of meshes were used for PN2ND and SN2ND calculations. For benchmark 1, hexahedron meshes of 4,352 and 15,625 elements were used, resulting in 19,079 and 24,362 vertices, respectively. A tetrahedral mesh of 11,812 elements (19,079 vertices) and hexahedral mesh of 5,240 elements (24,362 vertices) were used for benchmark 4. MOCFE calculations were performed with linear tetrahedral meshes upwards of 496,979 and 3,448,832 elements for benchmarks 1 and 4, respectively. The angular approximation varied from P_1 through P_{11} for PN2ND, and up to 288 discrete angles on the sphere were used for SN2ND and MOCFE. Tables 4 and 5 summarize the eigenvalue results for benchmark 1 and 4, respectively, showing the results of each solver in UNIC in addition to solutions from VARIANT. Although the eigenvalue results are somewhat scattered, the calculated control rod worths are within one standard deviation of the reference solutions, except for the full control rod worth of the fourth benchmark. The convergence behavior of the solutions indicated that more mesh refinement is necessary in all solvers of UNIC to converge to the reference solution. For example, Table 6 shows the eigenvalue solution of MOCFE for the mesh refinement study for the benchmark 4, where the angular cubature was fixed at 18 directions and a maximum trajectory area of 0.05cm^2 was used. An asymptotic convergence toward the reference solution is observable, but the last two meshes still result in nearly a 40 to 50 pcm change in the eigenvalue. It is also noted that MOCFE requires many more spatial elements than SN2ND for these homogenized problems, a typical by-product of the flat source approximation in MOC.

Assembly homogenized benchmark calculations were also performed for the Advanced Burner Test Reactor (ABTR) design [59,60]. Using the MC²-3 code, composition-dependent isotopic cross sections were generated in 9- and 33-group structures. The reference eigenvalue solutions obtained with 9- and 33-group MCNP calculations were 1.00813 ± 0.00004 and 1.01406 ± 0.00004 , respectively. PN2ND and SN2ND calculations were performed with various angular approximations and

Table 3. Comparison of Measured and Calculated Spectral Indices for LANL Assemblies

Assembly	Data		$\sigma_f^{U238} / \sigma_f^{U235}$	$\sigma_f^{Np237} / \sigma_f^{U235}$	$\sigma_f^{U233} / \sigma_f^{U235}$	$\sigma_f^{Pu239} / \sigma_f^{U235}$
GODIVA	Experiment		0.1643 ± 0.0018	0.8516 ± 0.013	1.59 ± 0.03	1.4152 ± 0.0250
	C/E	MCNP	0.960	0.975	0.987	0.977
		MC ² -3	0.958	0.974	0.987	0.977
JEZEBEL	Experiment		0.2133 ± 0.0023	0.9835 ± 0.016	1.578 ± 0.027	1.4609 ± 0.013
	C/E	MCNP	0.978	0.988	0.986	0.975
		MC ² -3	0.968	0.986	0.987	0.975
JEZEBEL -23	Experiment		0.2131 ± 0.0026	0.9970 ± 0.015		
	C/E	MCNP	0.989	0.984		
		MC ² -3	0.988	0.998		
FLATTOP-25	Experiment		0.1492 ± 0.0016	0.7804 ± 0.01	1.608 ± 0.003	1.3847 ± 0.012
	C/E	MCNP	0.968	0.988	0.975	0.982
		MC ² -3	0.966	0.988	0.975	0.982
FLATTOP-Pu	Experiment		0.1799 ± 0.002	0.8561 ± 0.012		
	C/E	MCNP	0.984	0.996		
		MC ² -3	0.970	0.992		
FLATTOP-23	Experiment		0.1916 ± 0.0021	0.9103 ± 0.013		
	C/E	MCNP	0.976	0.997		
		MC ² -3	0.976	0.998		

a) MCNP results are from Reference 55

b) MC²-3 rows denotes results from MC²-3/TWODANT calculation**Table 4.** Summary of Eigenvalue Results for Takeda Benchmark 1

	Control Rod In	Control Rod Out	Control Rod Worth
Reference	0.96240 $\pm$ 0.00071	0.97760 $\pm$ 0.00069	0.01616 $\pm$ 0.00099
VARIANT	0.96240	0.97699	0.01552
PN2ND	0.96239	0.97649	0.01501
SN2ND	0.96245	0.97717	0.01565
MOCFE	0.96171	0.97686	0.01613

hexahedral meshes, one of which is shown in Fig. 13. PN2ND and SN2ND eigenvalue solutions are summarized in Tables 7 and 8, respectively. The asymptotic convergence toward the reference solution is clearly seen for both PN2ND and SN2ND solutions, but a residual of ~ 50 pcm is observable in each of the two codes. A similar residual was observed in the VARIANT solution obtained with a P_7 angular approximation and a spatial approximation of eighth-, fourth- and third-order polynomials for flux, source, and leakage, respectively, which are usually sufficient for most problems. The residual error is associated with the leakage and source approximation in addition to the angular variable (primary source). In previous work [61], it has

Table 5. Summary of Eigenvalue Results for Takeda Benchmark 4

	Control Rod In	Control Rod Half	Control Rod Out
Reference	0.88001 ± 0.00038	0.98340 ± 0.00039	1.09515 ± 0.00040
VARIANT	0.88009	0.98415	1.09636
PN2ND	0.87960	0.98365	1.09599
SN2ND	0.87877	0.98275	1.09494
MOCFE	0.87796	0.98164	1.09353
	Half Control Rod Worth		Full Control Rod Worth
Reference	0.10376 ± 0.00056		0.22323 ± 0.00054
VARIANT	0.10400		0.22414
PN2ND	0.10420		0.22446
SN2ND	0.10426		0.22466
MOCFE	0.10423		0.22453

Table 6. Spatial Refinement Results of MOCFE for Takeda Benchmark 4

Number of Elements	Control Rod In	Control Rod Half	Control Rod Out
85,538	1496	1472	1474
194,833	919	872	854
386,650	662	596	575
897,248	426	346	329
2,113,902	310	223	209
3,448,832	261	176	162

* Errors in pcm from reference solutions

been shown that all such errors in VARIANT can be removed by full space-angle refinement.

The finite element methodologies in PN2ND and SN2ND have made space-angle refinement much more difficult. The SN2ND results indicate angular convergence using the S_{10} Carlson even-moment cubature, but clearly residual error is still present. Unlike the VARIANT code, with the finite element methodology, the mesh must be carefully refined near the steep gradients observed in the even-parity flux solution (or the large discontinuities observed in the odd-parity flux solution). Such mesh refinement is tedious and typically leads to many more spatial degrees of freedom than an equivalent VARIANT

Table 7. PN2ND Eigenvalue Solutions for ABTR Benchmark

Angular Order	Spatial Mesh Approximation (vertices)		
	78243	113873	461219
P ₁	-1660	-1655	-1594
P ₃	-366	-359	-249
P ₅	-254	-246	-109
P ₇	-228	-219	-70
P ₉	-220	-211	-55
P ₁₁	-217		

Table 8. SN2ND Eigenvalue Solutions for ABTR Benchmark

Angular Directions	Spatial Mesh Approximation (vertices)				
	78243	113873	461219	671219	785801
32	-241	-233	-69	-64	-59
50	-220	-210	-47	-40	-37
72	-225	-217	-51		
98	-216	-207	-43		
288	-216

Fig. 13. Hexahedron Mesh Used for ABTR Benchmark

Table 9. Multiplication Factors of ZPPR-15A Experiments

Loading	Experiment ^{a)}	VIM	MC ² -2/SDX/DIF3D	MC ² -3/DIF3D
15	1.00046	0.99985 ± 0.00020	0.99593 (-453)	0.99740 (-306)
16	0.99627	0.99571 ± 0.00021	0.99178 (-449)	0.99327 (-300)
20	0.99853	0.99742 ± 0.00019	0.99426 (-427)	0.99550 (-303)

^{a)} A maximum uncertainty of ~180 pcm was estimated for the experimental values.

^{b)} Numbers in parenthesis indicate pcm difference from experimental values.

Fig. 14. Geometrical Layout of ZPPR-15A (Octant Core)

calculation. Combined with the advantages of the structured geometry treatment over an unstructured one, the PN2ND and SN2ND solvers require more computational time on assembly homogenized problems, and in many cases can require several thousand processors to produce the same solution in a comparable time. From this study, it is concluded that the finite element-based SN2ND and PN2ND solvers are not the most efficient methodologies for treating assembly homogenized problems, especially when compared with a traditional nodal approach. However, nodal formulations typically become less attractive for smaller mesh sizes and require error-prone homogenization and dehomogenization techniques, which complicate multiphysics coupling.

3.4 ZPR-6 and ZPPR Critical Experiments

Combined performance of MC²-3 and UNIC was tested against the ZPR-6 and ZPPR critical experiments. The one-dimensional ultrafine group transport capability of MC²-3

Table 10. PN2ND Multiplication Factors for ZPPR-15A Loading 15 (ENDF/B-V.2 Data)

Angular Flux Expansion Order	Scattering Order	Multiplication Factor
P ₁	P ₁	0.99258
P ₃	P ₃	0.99640
P ₅	P ₅	0.99651
VIM Monte Carlo		0.99616 ± 0.00010

was first tested by analyzing the three loadings of ZPPR-15A and comparing the results to those of a previous study [50]. Figure 14 shows the geometrical layout for the ZPPR-15A Loading 15. Cell-averaged drawer cross sections were generated based on the ENDF/B-VII.0 data in a 230-group structure using the 1-D transport capability of MC²-3, while in the previous study they were generated based on the ENDF/V.2 and ENDF/VII.0 data using the MC²-2 and SDX codes. To compare with the previous results, the core calculations were performed with the DIF3D nodal transport option [62], whose accuracy is comparable to S₄ discrete ordinate solutions with P₁ scattering approximation. It is noted that the VARIANT option cannot be used for these calculations because the ~5 cm node leads to convergence problems. (While these issues can be fixed, it is not the focus of our development.)

Table 9 compares the calculated multiplication factors with the experimental values and VIM Monte Carlo solutions. The VIM Monte Carlo solutions were obtained using the as-built plate-by-plate models. It can be seen that MC²-3 cross sections improve the DIF3D solutions by ~150 pcm relative to MC²-2/SDX cross sections. However, MC²-3/DIF3D solutions still underestimate the multiplication factor by ~300 pcm Δk compared to VIM. A previous, unpublished study with the PN2ND solver and ENDF/B-V.2 data showed that the multiplication factor

of the ZPPR-15A Loading 15 almost converged using a P_3 angular flux approximation and a P_3 scattering order as shown in Table 10. It can be seen that the multiplication factor increases by ~ 400 pcm when the angular approximation order for flux and scattering is increased to P_3 from P_1 . Thus, it is expected that the calculated multiplication factor would be close to the experimental value, when the higher order angular approximations are applied.

To validate the accuracy of the SN2ND solver in addition to the cross section generation algorithms of MC²-3, the uranium-fueled ZPR-6/6A and the high Pu-240 core of ZPR-6/7 that includes a high Pu-240 zone at the center of the plutonium-fueled core were analyzed. The ZPR-6/6A assembly was slightly simplified by modeling similar drawers by a representative one, whereas the as-built models were used for ZPR-6/7 by modeling more than 100 drawer types explicitly. Cell-averaged drawer cross sections were generated based on the ENDF/B-VII.0 data in 9-, 33-, 70-, 116-, and 230-group structures using the 1-D transport capability of MC²-3. Several meshes and angular cubatures were implemented at each energy resolution.

For ZPR-6/6A, it was determined that a mesh with about 470,000 vertices and an S_6 angular cubature provided sufficient space-angle convergence (within 10 pcm). The reference continuous-energy VIM Monte Carlo solution of the plate-by-plate heterogeneous problem was 0.99981 ± 0.00025 . The SN2ND solutions obtained with 9-, 33-, 116-, and 230-energy groups showed eigenvalue errors of 26, -15, -16, and -15 pcm, respectively. That is, after an initial adjustment of 41 pcm from 9- to 33-energy groups, no further changes are observed in as many as 230 groups. Since these eigenvalues were generated with full space-angle convergence (496,496 quadratic vertices and 48

discrete ordinates), these results successfully indicate the accuracy of MC²-3 and SN2ND.

Table 11 summarizes the SN2ND eigenvalue results for the loading 104 of ZPR-6/7. The results were obtained using the PN2ND solver in diffusion theory (with transport cross sections provided by MC²-3). The experimental value was 1.00072 ± 0.00085 , where the uncertainty was estimated by adding the measurement uncertainty of 4.6 pcm to the combined uncertainty of 80.4 pcm for geometry and composition uncertainties. The results in Table 11 show the convergence of SN2ND solutions with respect to energy, mesh, and angular cubature, along with the importance of anisotropic scattering. Hexahedral meshes were used for all of these calculations where the 57,132 vertex mesh is considered a test case since it uses linear hexahedral trial functions. The 223,928 vertex mesh assumes ~ 8 cm axial element sizes and one hexahedron per homogenized drawer. The 805,185 vertex mesh also uses ~ 8 cm axial element sizes but it defines 4 hexahedrons per homogenized drawer, which yields about 9 pcm compared with the 223,928 vertex mesh. The 1,467,429 vertex mesh also uses 4 hexahedrons per homogenized drawer but targets a ~ 5 cm axial mesh size which yields no significant improvement over the 805,185 vertex mesh. In general, the 805,185 vertex mesh is considered sufficiently refined for our needs. The angular cubature has virtually no impact on this homogeneous core, but a ~ 40 pcm error associated with anisotropic scattering occurs. No attempt was made to use P_5 anisotropic scattering, but experience indicates that not much difference exists between P_3 and P_5 scattering on these types of problems. The convergence with respect to energy is quite curious because the 70- and 230-group solutions appear identical while the 116-group result is not consistent. This same behavior was not observed in the loading 106 results shown below, but it is worth investigating in the future.

Table 11. SN2ND and Diffusion Multiplication Factors for ZPR-6/7 Loading 104

Mesh Vertices	Angular Directions	Scattering Order	Number of Energy Groups				
			9	33	70	116	230
57,132	32	P_1	0.99676	0.99828	0.99905	0.99880	0.99905
223,928	32	P_1	0.99845	1.00001	1.00080	1.00055	1.00081
805,185	32	P_1	0.99854	1.00010	1.00089	1.00064	1.00090
805,185	32	P_3	0.99888	1.00044	1.00123		
805,185	72	P_1	0.99853	1.00009	1.00088	1.00063	1.00089
805,185	72	P_3	0.99887	1.00044	1.00122		
1,467,429	32	P_1	0.99855	1.00011	1.00090	1.00065	1.00092
1,467,429	Diffusion	P_0	0.99911	1.00065	1.00145	1.00105	1.00130

* Experimental value = 1.00072 ± 0.00085

The reaction rate distributions measured with activation foils were also calculated by superimposing the local flux shapes from MC²-3 to the global flux distributions from SN2ND. Figure 15 shows a preliminary result for the normalized radial distributions of enriched uranium (EU) fission, depleted uranium (DU) capture and fission, and Pu-239 fission reactions. All data are arbitrarily normalized to unity at the 66.2 cm radial position. The maximum deviation of the calculated values (C in Fig. 15) from the experimental values (E in Fig. 15) was 2.7 % for EU fission, 1.7% for DU capture, 3.5% for DU fission, and 3.4% for Pu-239 fission, while the estimated measurement uncertainties were 1.6, 2.0, 2.8, and 1.5%, respectively.

Table 12 summarizes the SN2ND eigenvalue results for the loading 106 of ZPR-6/7. The experimental value was 1.00091 ± 0.00086 , where the uncertainty was determined by adding the measurement uncertainty of 5.8 pcm to the combined uncertainty of 80.4 pcm for geometry and composition uncertainties. The continuous-energy Monte Carlo solution obtained from MCNP5 was 1.00049 ± 0.00007 . Again, it can be seen that the SN2ND solution converges with refined energy, mesh, angular cubature and scattering order. Figure 16 shows some selected flux plots from the 70-group calculation of the loading 106 of ZPR-6/7. As shown in Fig. 10, beryllium oxide was loaded in the four drawers surrounding the sodium drawer at the core center to enhance the worth of the simulated B₄C control rod. This causes significant localized flux changes with respect to those drawers without beryllium. For each group plot in Fig. 16, the right hand picture displays the face of the movable matrix half as displayed in Fig. 10, while the left hand picture shows the flux solution for the

active core of the stationary side (everything inside the depleted uranium blanket). For the stationary side, the high Pu-240 zone is separated to display the axial flux solution along with the central BeO modified drawers, allowing the axial flux variation to be observed. As can be seen, the BeO drawers yield only modest impacts on the high-energy and unresolved resonance regions. In the epithermal range, a substantial peak is observed in the flux solution due to the improved scattering source derived from the BeO drawer. To validate the calculated flux

Fig. 15. Radial Distributions of Reaction Rates for Loading 104 of ZPR-6/7

Table 12. SN2ND Multiplication Factors for ZPR-6/7 Loading 106

Mesh Vertices	Angular Directions	Scattering Order	Number of Energy Groups				
			9	33	70	116	230
103,684	32	P ₁	0.99739	0.99920	0.99954	0.99973	
299,276	32	P ₁	0.99851	1.00037	1.00071	1.00091	1.00123
408,112	32	P ₁	0.99852	1.00037	1.00071	1.00092	1.00123
408,112	32	P ₃	0.99887	1.00072	1.00105	1.00126	1.00157
408,112	72	P ₁	0.99851	1.00036	1.00070	1.00091	1.00122
408,112	72	P ₃	0.99886	1.00070	1.00105	1.00125	
1,001,437	32	P ₁	0.99822	1.00006	1.00040	1.00060	1.00091
1,001,437	32	P ₃	0.99856	1.00039	1.00074	1.00093	
1,001,437	72	P ₁	0.99822	1.00006	1.00040	1.00060	1.00091
1,001,437	72	P ₃	0.99856	1.00040		1.00094	

* Experimental value = 1.00091 ± 0.00086 ; MCNP5 result = 1.00049 ± 0.00007

Fig. 16. Selected Flux Plots for Loading 106 of ZPR-6/7

distribution, the reaction rate distributions measured with activation foils are being analyzed.

3.5 Parallel Computation Performance

All three solvers of UNIC can run on small- and large-scale parallel machines. The PN2ND, MOCFE, and SN2ND solvers have been executed on thousands, tens of thousands, and hundreds of thousands of processors, respectively, but the parallel performances of PN2ND and MOCFE have not been studied systematically. Thus, this subsection focuses on the parallel performance of the SN2ND solver, which recently merited finalist consideration for the 2009 Gordon Bell Supercomputing Prize [63]. This parallel performance study was performed using a plate-by-plate model of ZPR-6/6A. Since the goal was to investigate parallel performance, drawer-averaged cross sections were used rather than regenerating plate-wise cross sections for all drawer types included in the assembly.

With regard to parallel calculations, two measures of performance are typically considered: strong and weak scalabilities. Strong scalability shows how well time-to-solution can be reduced by increasing the number of processors. Weak scalability shows how efficiently the methodology can solve ever larger problems with proportionally larger processor counts. The weak and

strong scalabilities of SN2ND were studied on three high-performance computers. Two of the machines are the IBM BlueGene/P architectures hosted by ANL (40 racks with a total of 163,840 cores) [37] and by Jülich Supercomputing Center (JSC) in Germany (72 racks with a total of 294,912 cores) [64]. The third machine is the Cray XT5 hosted by Oak Ridge National Laboratory (225,000 cores) [65].

The Oak Ridge machine has the fastest individual processors and highest memory per core of the three machines, but the network speed on the IBM machine is notably more efficient for our calculations. Because of the low memory capabilities of the IBM machines, the scalability studies were generally limited to 9 or 33 energy groups, although larger group calculations were performed using as many as 116 groups but with coarser space-angle representations. Because UNIC can handle parallelization by space, angle, and energy, strong and weak scalabilities can be considered for each variable both separately and together, but this analysis was restricted to parallelization by space and angle separately.

Table 13 shows the strong scalability results in space on Blue Gene/P. A 15 million vertex mesh was used with 1.8 million hexahedral elements, a 9-group cross-section set, and an S_6 angular cubature. Because of the memory

limitations on BlueGene/P (512 MB per core), it was difficult to get a large number of vertices per core, and thus this study was performed by spreading the mesh over 4,096 cores yielding ~7,324 vertices per core. The mesh was further partitioned over 8 times as many processors leading to ~1,831 vertices per process. Below 800 vertices per process, extremely diminished performance was typically seen due to the increased communication and iteration requirements. This type of behavior is not unusual for the SSOR preconditioner used for the conjugate gradient solver in PETSc (where the iteration count increases as the space gets partitioned over more subdomains), and thus it is not necessarily indicative of the parallel machine or the SN2ND methodology. Note that the set of angular directions is spread into two sets of 4,096 cores. Each such set has the mesh replicated, thus giving a hierarchical division of work.

Because the synthetic diffusion equation has not been partitioned over the full space-angle set of processors yet, the strong scalability in angle (<75%) is not as good as that in space. Future development is focused on distributing the diffusion system and implementing spatial multigrid methods such that the number of vertices per core can be increased and the memory problems currently being experienced can be reduced. Spatial multigrid, when effective, has been shown to maintain the high scalability shown above when applied to similar equations to those solved by SN2ND.

Because space-angle convergence studies often necessitate increasing space-angle resolutions until a desired accuracy is achieved, weak scaling studies and performance are more important than strong scalability. In fact, several of the ZPR-6 calculations employed the maximum number of processors and memory available on the parallel machines, despite the fact that none of these studies had indicated full space-angle convergence (even without considering parallelization by energy). Table 14 gives the weak scaling performance for calculation with 9 energy groups on Blue Gene/P. As can be seen, the performance is excellent up to 131,072 cores, but beyond this a rapid drop-off is observable. An inspection of the timing history indicates that this is mostly attributable to a breakdown in performance of the “global reduce” operation associated with the within-group scattering source update. This operation on user-defined communicators is not expected to be implemented optimally in vendor-supplied MPI libraries. Currently, various approaches to mapping MPI processes to cores are being explored to obtain better performance for high core count cases.

Table 15 gives the weak angle scalability of SN2ND on XT5 for a calculation with 33 energy groups. Similar to the BlueGene/P results, it shows an observable drop in performance, but this time it occurs at 100,000 cores, which is associated with the different network characteristics of the two machines. However, it is important to note that the faster processor speed and larger memory per core of the

Table 13. Strong Spatial Scalability of SN2ND on BlueGene/P at ANL

Total Cores	Vertices Per Process	Total Time (sec)	Strong Scaling
8,192	7,324	2,402	100%
16,384	3,662	1,312	92%
24,576	2,441	873	92%
32,768	1,831	637	94%

Table 14. Weak Angle Scalability of SN2ND on BlueGene/P (Combined ANL and JSC)

Total Cores	Angular Directions	Total Time (sec)	Weak Scaling
32,768	32	579	100%
73,728	72	572	101%
131,072	128	581	100%
163,840	160	691	84%
294,912	288	763	76%

Table 15. Weak Angle Scalability of SN2ND on XT5 (ORNL)

Total Cores	Angular Directions	Total Time (sec)	Weak Scaling
16,512	32	1891	100%
37,152	72	1901	99%
66,048	128	1829	103%
103,200	200	2050	92%
148,608	288	2298	82%
222,912	432	2517	75%

XT5 machines allows the 33-group problem to be solved in a comparable amount of time to the 9-group calculation on Blue Gene/P. Using XT5, it was possible to obtain ZPR-6/6A results containing over 0.9 trillion degrees of freedom.

For spatial weak scaling, the previous calculations with assembly-homogenized problems indicated weak scalability between 70 and 80%, but these numbers do not extend past a few thousand processors. For the plate-by-plate heterogeneous problems used in this study, the required degree of mesh refinement is somewhat fixed because of the physics in the problem (optical thickness of regions). Unlike the assembly-homogenized problems, mesh refinement in heterogeneous problems can easily change the condition number of the coefficient matrix and thus pollute the timing data. As an example, finite elements can be defined with very bad aspect ratios that yield a poor coefficient matrix condition number. More vertices introduced with mesh refinement would typically translate to a decreasing weak scalability, but in our case with axial mesh refinement, it generally improves the condition number of the coefficient system so that it appears to obtain greater than 100% weak scaling. However, if the mesh is artificially refined past the point of asymptotic convergence, the condition number becomes worse and it appears to have very poor weak scaling. Neither of these scenarios tells much about the actual weak scaling performance of the coding. In the future, it will be necessary to devise contrived heterogeneous benchmark problems that retain the same physics properties but allow the domain size to be increased in some regular fashion.

4. CONCLUSION

As part of U.S. DOE's NEAMS program, ANL is developing a suite of modern fast reactor simulation tools. The general goal is to reduce the uncertainties and biases in various areas of reactor design activities by providing enhanced prediction capabilities. Under this fast reactor simulation program, a high-fidelity deterministic neutron transport code UNIC and a multigroup cross section generation code MC²-3 are being developed. The end goal of this development is to produce an integrated neutronics code that enables the high-fidelity description of a nuclear reactor and simplifies the multistep design process by direct and accurate coupling with thermal hydraulics and structural mechanics calculations.

The application scope targeted for UNIC ranges from the homogenized assembly approaches prevalent in current reactor analysis methodologies to explicit geometry, time-dependent transport calculations that are directly coupled to thermal-hydraulics and structural mechanics calculations in reactor accident simulations. Three transport equation solvers have been developed based upon the unstructured finite-element mesh representation: PN2ND, SN2ND, and MOCFE. PN2ND and SN2ND solve the second-order even-parity transport equation with spherical harmonics and discrete ordinates angular approximations, respectively. MOCFE is a method of characteristics solver for the first-order transport equation, of which targeted application is

for explicit geometry, fine-group problems. The nodal transport code VARIANT is also being rebuilt in UNIC as NODAL which is focused on solving modern assembly homogenized problems. To generate multigroup cross sections in fast reactor applications, MC²-3 has been developed by improving the resonance self-shielding and spectrum calculation methods of MC²-2, and by integrating the one-dimensional cell calculation capabilities of SDX.

The solution methods and modeling capabilities of the three transport solvers of UNIC were briefly discussed. The finite element discretization of spatial domain and the spherical harmonics and discrete ordinates angular approximations implemented in UNIC solvers were discussed along with the iterative solution methods for the resulting system of discretized equations. The space-angle-energy domain decomposition approach for massively parallel computations was also discussed. It is noted that SN2ND can already be used on both the world's largest and the world's fastest machines (Top 500 in November 2009) using space-angle partitioning alone. The slowing-down calculation methods of MC²-3 were also discussed, focusing on enhanced methods for resonance self-shielding and spectrum calculations.

The performances of MC²-3 and UNIC were tested using several well known benchmark problems and fast critical experiments. The MC²-3 results for homogeneous medium and unit cell calculations and the transport solutions obtained with MC²-3 cross sections for simplified benchmark problems derived from fast critical experiments were compared to MCNP and VIM Monte Carlo solutions. The results indicated that MC²-3 produces accurate multigroup cross sections for homogeneous media and heterogeneous unit cells. The UNIC calculations for three-dimensional homogenized benchmark problems with given sets of group cross sections showed that for all three solvers, asymptotic convergence toward the reference multigroup Monte Carlo solutions was observable. This paper also discussed the fact that for assembly-homogenized problems, the performance of nodal spatial discretization is quite superior in performance and accuracy to a low order finite element method. Combined tests of MC²-3 and UNIC for the ZPR-6 and ZPPR critical experiments also showed that MC²-3/SN2ND calculations are sufficiently accurate for analyzing these experiments. Some timing results for the SN2ND solver on the ZPR-6 critical assembly benchmarks were also discussed. While they can only give a snapshot of the solver performance in this phase of its development, these numbers indicate successful parallel performance, achieving 76% weak angle scaling on 294,912 cores and 94% strong spatial scaling on 32,768 cores.

Most of the UNIC calculations were performed on high performance computers unavailable to most reactor designers, and substantial research and development is still required to make UNIC applicable to practical design calculations. The most important research that remains is to develop a better preconditioner on the space-angle system

along with one on the entire space-angle-energy system, in particular for MOCFE. Other improvement needs for each solver of UNIC were discussed in Sections 2 and 3 while discussing the solution methods and initial test results. Our current efforts are focused on developing a multigrid preconditioner in SN2ND and a HFG transport calculation capability with anisotropic scattering sources in MC²-3. We also are integrating the MOCFE solver into the MC²-3 code to improve the accuracy of the cross section generation, and we are building a nodal transport solver in UNIC for application to assembly homogenized problems.

Our future efforts will include carrying out time-dependent multi-physics calculations for full reactor core-sized problems on current and future large-scale architectures, simulating various accident scenarios such as control rod ejections and loss of flow without scram. The higher fidelity solution capability will not only allow us to better understand the behavior of existing reactor systems, but also predict the behavior of many advanced design concepts.

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